

# Finite Volume Methods for Hyperbolic Problems

## Introduction to Finite Volume Methods

- Comparison to finite differences
- Conservation form, importance for shocks
- Godunov's method, wave propagation view
- Upwind for advection
- REA Algorithm
- Godunov applied to acoustics

# Finite difference method

Based on point-wise approximations:

$$Q_i^n \approx q(x_i, t_n), \quad \text{with } x_i = i\Delta x, \quad t_n = n\Delta t.$$

Approximate derivatives by finite differences.

**Ex:** Upwind method for advection equation if  $u > 0$ :

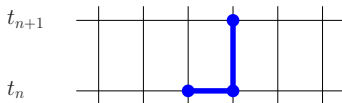
$$q_t + uq_x = 0$$

$$\frac{Q_i^{n+1} - Q_i^n}{\Delta t} + u \left( \frac{Q_i^n - Q_{i-1}^n}{\Delta x} \right) = 0$$

or

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} u (Q_i^n - Q_{i-1}^n).$$

Stencil:



# Finite differences vs. finite volumes

## Finite difference Methods

- Pointwise values  $Q_i^n \approx q(x_i, t_n)$
- Approximate derivatives by finite differences
- Assumes smoothness

## Finite volume Methods

- Approximate cell averages:  $Q_i^n \approx \frac{1}{\Delta x} \int_{x_{i-1/2}}^{x_{i+1/2}} q(x, t_n) dx$
- Integral form of conservation law,

$$\frac{\partial}{\partial t} \int_{x_{i-1/2}}^{x_{i+1/2}} q(x, t) dx = f(q(x_{i-1/2}, t)) - f(q(x_{i+1/2}, t))$$

leads to conservation law  $q_t + f_x = 0$  but also directly to numerical method.

# Finite volume method

$$Q_i^n \approx \frac{1}{h} \int_{x_{i-1/2}}^{x_{i+1/2}} q(x, t_n) dx$$

Integral form: 
$$\frac{\partial}{\partial t} \int_{x_{i-1/2}}^{x_{i+1/2}} q(x, t) dx = f(q(x_{i-1/2}, t)) - f(q(x_{i+1/2}, t))$$

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Integrate from  $t_n$  to  $t_{n+1} \implies$

$$\int q(x, t_{n+1}) dx = \int q(x, t_n) dx + \int_{t_n}^{t_{n+1}} (f(q(x_{i-1/2}, t)) - f(q(x_{i+1/2}, t))) dt$$

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Integrate from  $t_n$  to  $t_{n+1} \implies$

$$\int q(x, t_{n+1}) dx = \int q(x, t_n) dx + \int_{t_n}^{t_{n+1}} f(q(x_{i-1/2}, t)) - f(q(x_{i+1/2}, t)) dt$$

$$\begin{aligned} \frac{1}{\Delta x} \int q(x, t_{n+1}) dx &= \frac{1}{\Delta x} \int q(x, t_n) dx \\ &\quad - \frac{\Delta t}{\Delta x} \left( \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} f(q(x_{i+1/2}, t)) - f(q(x_{i-1/2}, t)) dt \right) \end{aligned}$$

Numerical method: 
$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} (F_{i+1/2}^n - F_{i-1/2}^n)$$

Numerical flux: 
$$F_{i-1/2}^n \approx \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} f(q(x_{i-1/2}, t)) dt.$$

# Upwind for advection as a finite volume method

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} (F_{i+1/2}^n - F_{i-1/2}^n)$$

$$F_{i-1/2} \approx \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} uq(x_{i-1/2}, t) dt.$$

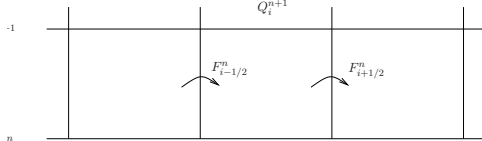
For  $u > 0$ :

$$F_{i-1/2}^n = uQ_{i-1}^n, \quad F_{i+1/2}^n = uQ_i^n$$

so

$$\begin{aligned} Q_i^{n+1} &= Q_i^n + \frac{\Delta t(uQ_{i-1}^n - uQ_i^n)}{\Delta x} \\ &= Q_i^n - \frac{\Delta tu}{\Delta x} (Q_i^n - Q_{i-1}^n) \end{aligned}$$

Stencil:  
( $x$ - $t$  plane)



# Upwind method for advection

Flux:  $f(q) = uq$

Numerical flux:  $F_{i-1/2}^n \approx \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} f(q(x_{i-1/2}, t)) dt.$

If  $q(x, t_n)$  is piecewise constant in each cell, then

$$F_{i-1/2}^n = \begin{cases} uQ_{i-1}^n & \text{if } u > 0, \\ uQ_i^n & \text{if } u < 0. \end{cases}$$



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This gives the **upwind method**:

$$Q_i^{n+1} = Q_i^n - \frac{u\Delta t}{\Delta x} (Q_i^n - Q_{i-1}^n) \quad \text{if } u > 0$$

$$Q_i^{n+1} = Q_i^n - \frac{u\Delta t}{\Delta x} (Q_{i+1}^n - Q_i^n) \quad \text{if } u < 0$$

# Conservation form

The method

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} (F_{i+1/2}^n - F_{i-1/2}^n)$$

is in **conservation form**.

The total mass is conserved up to fluxes at the boundaries:

$$\Delta x \sum_i Q_i^{n+1} = \Delta x \sum_i Q_i^n - \frac{\Delta t}{\Delta x} (F_{+\infty} - F_{-\infty}).$$

Note: an isolated shock must travel at the right speed!

# Nonlinear scalar conservation laws

Burgers' equation:  $u_t + \left(\frac{1}{2}u^2\right)_x = 0$ .

Quasilinear form:  $u_t + uu_x = 0$ .

These are equivalent for **smooth** solutions, not for shocks!

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Upwind methods for  $u > 0$ :

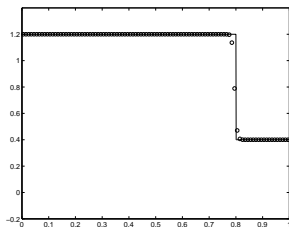
Conservative:  $U_i^{n+1} = U_i^n - \frac{\Delta t}{\Delta x} \left(\frac{1}{2}((U_i^n)^2 - (U_{i-1}^n)^2)\right)$

Quasilinear:  $U_i^{n+1} = U_i^n - \frac{\Delta t}{\Delta x} U_i^n (U_i^n - U_{i-1}^n)$ .

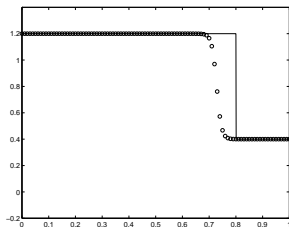
Ok for smooth solutions, not for shocks!

# Importance of conservation form

Solution to Burgers' equation using conservative upwind:



Solution to Burgers' equation using quasilinear upwind:



# Weak solutions depend on the conservation law

The conservation laws

$$u_t + \left(\frac{1}{2}u^2\right)_x = 0$$

and

$$(u^2)_t + \left(\frac{2}{3}u^3\right)_x = 0 \quad \text{i.e.} \quad q = u^2, \quad f(q) = \frac{2}{3}q^{3/2}$$

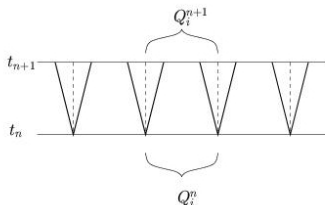
both have the same quasilinear form

$$u_t + uu_x = 0$$

but have different weak solutions,

different shock speeds!

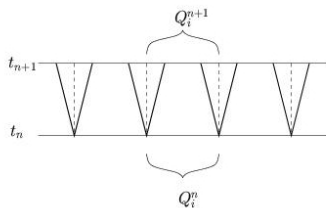
# Godunov's Method for $q_t + f(q)_x = 0$



1. Solve Riemann problems at all interfaces, yielding waves  $\mathcal{W}_{i-1/2}^p$  and speeds  $s_{i-1/2}^p$ , for  $p = 1, 2, \dots, m$ .

**Riemann problem:** Original equation with piecewise constant data.

# Godunov's Method for $q_t + f(q)_x = 0$

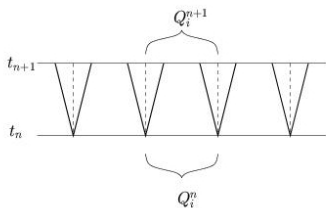


Then either:

1. Compute new cell averages by integrating over cell at  $t_{n+1}$ ,



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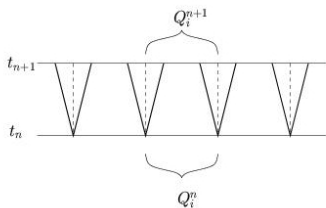


Then either:

1. Compute new cell averages by integrating over cell at  $t_{n+1}$ ,
2. Compute fluxes at interfaces and flux-difference:

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [F_{i+1/2}^n - F_{i-1/2}^n]$$

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2. Compute fluxes at interfaces and flux-difference:

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [F_{i+1/2}^n - F_{i-1/2}^n]$$

3. Update cell averages by contributions from all waves entering cell:

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [\mathcal{A}^+ \Delta Q_{i-1/2} + \mathcal{A}^- \Delta Q_{i+1/2}]$$

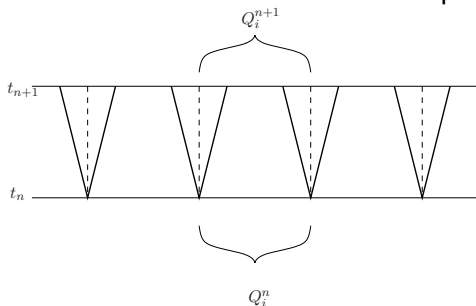
where  $\mathcal{A}^\pm \Delta Q_{i-1/2} = \sum_{i=1}^m (s_{i-1/2}^p)^\pm \mathcal{W}_{i-1/2}^p$ .

# Godunov's method with flux differencing

$Q_i^n$  defines a piecewise constant function

$$\tilde{q}^n(x, t_n) = Q_i^n \quad \text{for } x_{i-1/2} < x < x_{i+1/2}$$

Discontinuities at cell interfaces  $\implies$  Riemann problems.



$$\tilde{q}^n(x_{i-1/2}, t) \equiv q^\downarrow(Q_{i-1}, Q_i) \quad \text{for } t > t_n.$$

$$F_{i-1/2}^n = \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} f(q^\downarrow(Q_{i-1}^n, Q_i^n)) dt = f(q^\downarrow(Q_{i-1}^n, Q_i^n)).$$

# Upwind method for advection

Flux:  $f(q) = uq$

Numerical flux:  $F_{i-1/2}^n \approx \frac{1}{\Delta t} \int_{t_n}^{t_{n+1}} f(q(x_{i-1/2}, t)) dt.$

If  $q(x, t_n)$  is piecewise constant in each cell, then

$$F_{i-1/2}^n = \begin{cases} uQ_{i-1}^n & \text{if } u > 0, \\ uQ_i^n & \text{if } u < 0. \end{cases}$$

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This gives the **upwind method**:

$$Q_i^{n+1} = Q_i^n - \frac{u\Delta t}{\Delta x} (Q_i^n - Q_{i-1}^n) \quad \text{if } u > 0$$

$$Q_i^{n+1} = Q_i^n - \frac{u\Delta t}{\Delta x} (Q_{i+1}^n - Q_i^n) \quad \text{if } u < 0$$

# First-order REA Algorithm

- 1 **Reconstruct** a piecewise constant function  $\tilde{q}^n(x, t_n)$  defined for all  $x$ , from the cell averages  $Q_i^n$ .

$$\tilde{q}^n(x, t_n) = Q_i^n \quad \text{for all } x \in C_i.$$

- 2 **Evolve** the hyperbolic equation exactly (or approximately) with this initial data to obtain  $\tilde{q}^n(x, t_{n+1})$  a time  $\Delta t$  later.

- 3 **Average** this function over each grid cell to obtain new cell averages

$$Q_i^{n+1} = \frac{1}{\Delta x} \int_{C_i} \tilde{q}^n(x, t_{n+1}) dx.$$

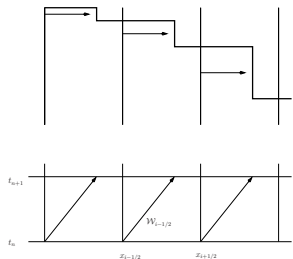
# Godunov's method for advection

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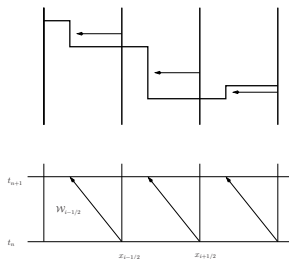
$$\tilde{q}^n(x, t_n) = Q_i^n \text{ for } x_{i-1/2} < x < x_{i+1/2}$$

Discontinuities at cell interfaces  $\implies$  Riemann problems.

$u > 0$

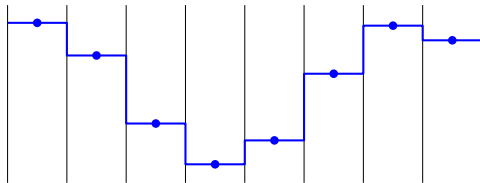


$u < 0$

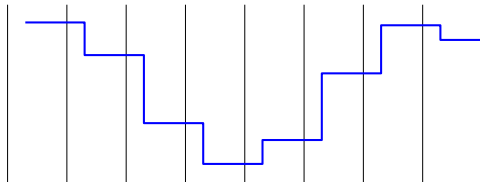


# First-order REA Algorithm

Cell averages and piecewise constant reconstruction:

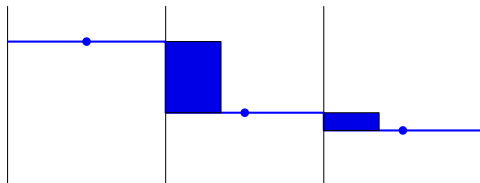


After evolution:





# Cell update



The cell average is modified by

$$\frac{u\Delta t \cdot (Q_{i-1}^n - Q_i^n)}{\Delta x}$$

So we obtain the upwind method

$$Q_i^{n+1} = Q_i^n - \frac{u\Delta t}{\Delta x} (Q_i^n - Q_{i-1}^n).$$

# Wave propagation form of cell update

The cell average is modified by

$$\frac{u\Delta t \cdot (Q_{i-1}^n - Q_i^n)}{\Delta x} = -\frac{\Delta t}{\Delta x} s \mathcal{W}_{i-1/2}$$

where  $\mathcal{W}_{i-1/2} = (Q_i^n - Q_{i-1}^n)$  is the wave strength and  $s = u$  is the wave speed.

The general upwind method for  $u < 0$  or  $u > 0$ :

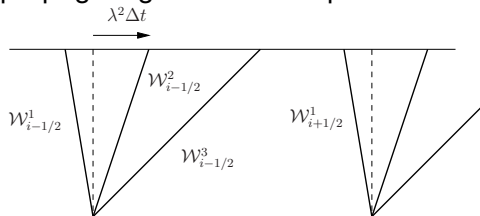
$$\begin{aligned} Q_i^{n+1} &= Q_i^n - \frac{\Delta t}{\Delta x} [u^+(Q_i^n - Q_{i-1}^n) + u^-(Q_{i+1}^n - Q_i^n)] \\ &= \frac{\Delta t}{\Delta x} [s^+ \mathcal{W}_{i-1/2} + s^- \mathcal{W}_{i-1/2}] \end{aligned}$$

where  $u^+ = \max(u, 0)$ ,  $u^- = \min(u, 0)$ .

This is the **wave propagation form** of upwind.

# Wave-propagation viewpoint

For linear system  $q_t + Aq_x = 0$ , the Riemann solution consists of waves  $\mathcal{W}^p$  propagating at constant speed  $\lambda^p$ .



$$Q_i - Q_{i-1} = \sum_{p=1}^m \alpha_{i-1/2}^p r^p \equiv \sum_{p=1}^m \mathcal{W}_{i-1/2}^p.$$

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [\lambda^2 \mathcal{W}_{i-1/2}^2 + \lambda^3 \mathcal{W}_{i-1/2}^3 + \lambda^1 \mathcal{W}_{i+1/2}^1].$$

# Godunov (upwind) for a linear system

$q_t + Aq_x = 0$  where  $A = R\Lambda R^{-1}$ . Define the matrices

$$\Lambda^+ = \begin{bmatrix} (\lambda^1)^+ & & & \\ & (\lambda^2)^+ & & \\ & & \ddots & \\ & & & (\lambda^m)^+ \end{bmatrix}, \quad \Lambda^- = \begin{bmatrix} (\lambda^1)^- & & & \\ & (\lambda^2)^- & & \\ & & \ddots & \\ & & & (\lambda^m)^- \end{bmatrix}.$$

and

$$A^+ = R\Lambda^+R^{-1}, \quad \text{and} \quad A^- = R\Lambda^-R^{-1}.$$

Note:

$$A^+ + A^- = R(\Lambda^+ + \Lambda^-)R^{-1} = R\Lambda R^{-1} = A.$$

Then Godunov's method becomes

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [A^+(Q_i - Q_{i-1}) + A^-(Q_{i+1} - Q_i)].$$

# Matrix splitting for upwind method

For  $q_t + Aq_x = 0$ , the upwind method (Godunov) is:

$$\begin{aligned}Q_i^{n+1} &= Q_i^n + \frac{\Delta t}{\Delta x} \left[ \sum_{p=1}^m (\lambda^p)^+ \alpha_{i-1/2}^p r^p + \sum_{p=1}^m (\lambda^p)^- \alpha_{i+1/2}^p r^p \right] \\&= Q_i^n + \frac{\Delta t}{\Delta x} [A^+ \Delta Q_{i-1/2} + A^- \Delta Q_{i+1/2}] \\&= Q_i^n + \frac{\Delta t}{\Delta x} [A^+ (Q_i^n - Q_{i-1}^n) + A^- (Q_{i+1}^n - Q_i^n)]\end{aligned}$$

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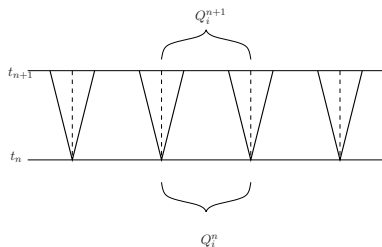
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Natural generalization of upwind to a system.

If all eigenvalues are positive, then  $A^+ = A$  and  $A^- = 0$ ,

If all eigenvalues are negative, then  $A^+ = 0$  and  $A^- = A$ .

# Godunov (upwind) on acoustics

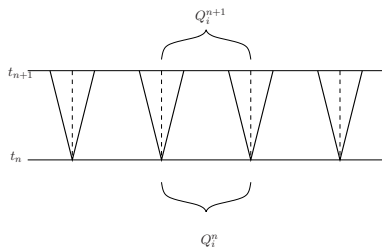


Data at time  $t_n$  :  $\tilde{q}^n(x, t_n) = Q_i^n$  for  $x_{i-1/2} < x < x_{i+1/2}$

Solving Riemann problems for small  $\Delta t$  gives solution:

$$\tilde{q}^n(x, t_{n+1}) = \begin{cases} Q_{i-1/2}^* & \text{if } x_{i-1/2} - c\Delta t < x < x_{i-1/2} + c\Delta t, \\ Q_i^n & \text{if } x_{i-1/2} + c\Delta t < x < x_{i+1/2} - c\Delta t, \\ Q_{i+1/2}^* & \text{if } x_{i+1/2} - c\Delta t < x < x_{i+1/2} + c\Delta t, \end{cases}$$

# Godunov (upwind) on acoustics



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So computing cell average gives:

$$Q_i^{n+1} = \frac{1}{\Delta x} \left[ c\Delta t Q_{i-1/2}^* + (\Delta x - 2c\Delta t) Q_i^n + c\Delta t Q_{i+1/2}^* \right].$$



# Godunov (upwind) on acoustics

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Solve Riemann problems:

$$Q_i^n - Q_{i-1}^n = \Delta Q_{i-1/2} = \mathcal{W}_{i-1/2}^1 + \mathcal{W}_{i-1/2}^2 = \alpha_{i-1/2}^1 r^1 + \alpha_{i-1/2}^2 r^2,$$

$$Q_{i+1}^n - Q_i^n = \Delta Q_{i+1/2} = \mathcal{W}_{i+1/2}^1 + \mathcal{W}_{i+1/2}^2 = \alpha_{i+1/2}^1 r^1 + \alpha_{i+1/2}^2 r^2,$$

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$$Q_{i+1}^n - Q_i^n = \Delta Q_{i+1/2} = \mathcal{W}_{i+1/2}^1 + \mathcal{W}_{i+1/2}^2 = \alpha_{i+1/2}^1 r^1 + \alpha_{i+1/2}^2 r^2,$$

The intermediate states are:

$$Q_{i-1/2}^* = Q_i^n - \mathcal{W}_{i-1/2}^2, \quad Q_{i+1/2}^* = Q_i^n + \mathcal{W}_{i+1/2}^1,$$

# Godunov (upwind) on acoustics

$$Q_i^{n+1} = \frac{1}{\Delta x} \left[ c\Delta t Q_{i-1/2}^* + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t Q_{i+1/2}^* \right].$$

Solve Riemann problems:

$$Q_i^n - Q_{i-1}^n = \Delta Q_{i-1/2} = \mathcal{W}_{i-1/2}^1 + \mathcal{W}_{i-1/2}^2 = \alpha_{i-1/2}^1 r^1 + \alpha_{i-1/2}^2 r^2,$$

$$Q_{i+1}^n - Q_i^n = \Delta Q_{i+1/2} = \mathcal{W}_{i+1/2}^1 + \mathcal{W}_{i+1/2}^2 = \alpha_{i+1/2}^1 r^1 + \alpha_{i+1/2}^2 r^2,$$

The intermediate states are:

$$Q_{i-1/2}^* = Q_i^n - \mathcal{W}_{i-1/2}^2, \quad Q_{i+1/2}^* = Q_i^n + \mathcal{W}_{i+1/2}^1,$$

So,

$$\begin{aligned} Q_i^{n+1} &= \frac{1}{\Delta x} \left[ c\Delta t(Q_i^n - \mathcal{W}_{i-1/2}^2) + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t(Q_i^n + \mathcal{W}_{i+1/2}^1) \right] \\ &= Q_i^n - \frac{c\Delta t}{\Delta x} \mathcal{W}_{i-1/2}^2 + \frac{c\Delta t}{\Delta x} \mathcal{W}_{i+1/2}^1. \end{aligned}$$

# Godunov (upwind) on acoustics

$$\begin{aligned}Q_i^{n+1} &= \frac{1}{\Delta x} \left[ c\Delta t Q_{i-1/2}^* + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t Q_{i+1/2}^* \right] \\&= \frac{1}{\Delta x} \left[ c\Delta t(Q_i^n - \mathcal{W}_{i-1/2}^2) + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t(Q_i^n + \mathcal{W}_{i+1/2}^1) \right] \\&= Q_i^n - \frac{c\Delta t}{\Delta x} \mathcal{W}_{i-1/2}^2 + \frac{c\Delta t}{\Delta x} \mathcal{W}_{i+1/2}^1 \\&= Q_i^n - \frac{\Delta t}{\Delta x} (c\mathcal{W}_{i-1/2}^2 + (-c)\mathcal{W}_{i+1/2}^1).\end{aligned}$$

# Godunov (upwind) on acoustics

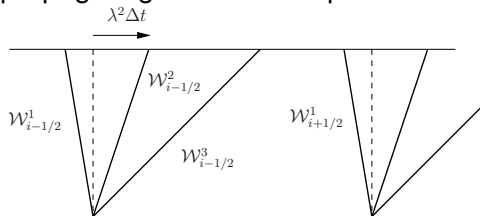
$$\begin{aligned}Q_i^{n+1} &= \frac{1}{\Delta x} \left[ c\Delta t Q_{i-1/2}^* + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t Q_{i+1/2}^* \right] \\&= \frac{1}{\Delta x} \left[ c\Delta t(Q_i^n - \mathcal{W}_{i-1/2}^2) + (\Delta x - 2c\Delta t)Q_i^n + c\Delta t(Q_i^n + \mathcal{W}_{i+1/2}^1) \right] \\&= Q_i^n - \frac{c\Delta t}{\Delta x} \mathcal{W}_{i-1/2}^2 + \frac{c\Delta t}{\Delta x} \mathcal{W}_{i+1/2}^1 \\&= Q_i^n - \frac{\Delta t}{\Delta x} (c\mathcal{W}_{i-1/2}^2 + (-c)\mathcal{W}_{i+1/2}^1).\end{aligned}$$

General form for linear system with  $m$  equations:

$$\begin{aligned}Q_i^{n+1} &= Q_i^n - \frac{\Delta t}{\Delta x} \left[ \sum_{p:\lambda^p > 0} \lambda^p \mathcal{W}_{i-1/2}^p + \sum_{p:\lambda^p < 0} \lambda^p \mathcal{W}_{i+1/2}^p \right] \\&= Q_i^n - \frac{\Delta t}{\Delta x} \left[ \sum_{m=1}^P (\lambda^p)^+ \mathcal{W}_{i-1/2}^p + \sum_{m=1}^P (\lambda^p)^- \mathcal{W}_{i+1/2}^p \right]\end{aligned}$$

# Wave-propagation viewpoint

For linear system  $q_t + Aq_x = 0$ , the Riemann solution consists of waves  $\mathcal{W}^p$  propagating at constant speed  $\lambda^p$ .



$$Q_i - Q_{i-1} = \sum_{p=1}^m \alpha_{i-1/2}^p r^p \equiv \sum_{p=1}^m \mathcal{W}_{i-1/2}^p.$$

$$Q_i^{n+1} = Q_i^n - \frac{\Delta t}{\Delta x} [\lambda^2 \mathcal{W}_{i-1/2}^2 + \lambda^3 \mathcal{W}_{i-1/2}^3 + \lambda^1 \mathcal{W}_{i+1/2}^1].$$