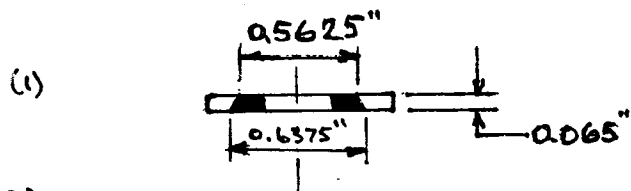


FRUSTUM 1: WASHER



(2)

$$E = 30 \text{ Mpsi}, t = 0.065$$

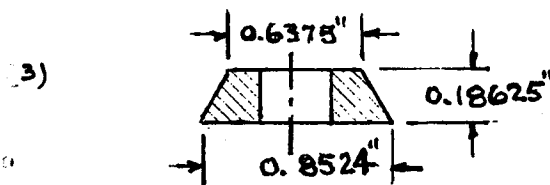
$$D = 0.5625 + 2(0.065) = 0.577$$

$$= 0.6375 \text{ in}$$

(2)

$$k = 78.58 \text{ Mlb/in, BY COMPUTER}$$

FRUSTUM 2: CAP PORTION



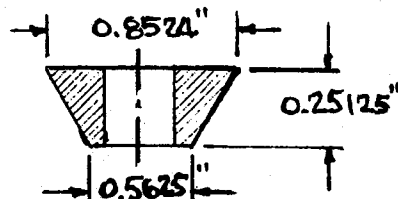
$$E = 14 \text{ Mpsi}, t = 0.18625 \text{ in}$$

$$D = 0.6375 + 2(0.18625) = 0.577$$

$$= 0.8524 \text{ in}$$

$$k = 23.43 \text{ Mlb/in, BY COMPUTER}$$

FRUSTUM 3: FRAME AND CAP



$$E = 14 \text{ Mpsi}, t = 0.25125 \text{ in}$$

$$k = 14.31 \text{ Mlb/in, BY COMPUTER}$$

$$k_m = \frac{1}{\frac{1}{78.58} + \frac{1}{23.43} + \frac{1}{14.31}} = 7.98 \text{ Mlb/in}$$

Ans.

FOR $h_{\text{TOTAL}} = 2(3/8) + (1/4) = 1 \text{ in}$
 SO IT IS THREADED ALL THE WAY.
 SINCE $A_t = 0.0775 \text{ in}^2$

$$k_b = \frac{0.0775(30)}{0.5025} = 4.63 \text{ Mlb/in}$$

Ans.

B-32 FIG. B-22. $t_1 = 0.25 \text{ in}$,
 $h = 0.25 + 0.065 = 0.315 \text{ in}$
 $L = h + d/2 = 0.315 + (3/16) = 0.5025 \text{ in}$
 $D_1 = 1.5(0.375) + 0.577(0.5025) = 0.8524 \text{ in}$
 $D_2 = 1.5(0.375) = 0.5625 \text{ in}$
 $L/2 = 0.5025/2 = 0.25125 \text{ in}$

B-33 (a) $F'_b = RF'_{bmax} \sin \theta$

Half the external moment is contributed by the line load in the interval

$$0 \leq \theta \leq \pi$$

$$\begin{aligned} \frac{M}{2} &= \int_0^{\pi} F'_b R^2 \sin \theta d\theta \\ &= \int_0^{\pi} F'_{bmax} R^2 \sin^2 \theta d\theta \end{aligned}$$

$$\frac{M}{2} = \frac{\pi}{2} F'_{bmax} R^2$$

from which $F'_{bmax} = \frac{M}{\pi R^2}$

$$\begin{aligned} F_{max} &= \int_{\phi_1}^{\phi_2} F'_b R \sin \theta d\theta \\ &= \frac{M}{\pi R^2} \int_{\phi_1}^{\phi_2} R \sin \theta d\theta \\ &= \frac{M}{\pi R} (\cos \phi_1 - \cos \phi_2) \end{aligned}$$

Noting $\phi_1 = 75^\circ$, $\phi_2 = 105^\circ$

$$\begin{aligned} F_{max} &= \frac{12\ 000}{\pi(8/2)} (\cos 75^\circ - \cos 105^\circ) \\ &= 494\ \text{lb} \quad \underline{\text{Ans.}} \end{aligned}$$

(b)

$$\begin{aligned} F_{max} &= F'_{bmax} R \Delta\phi = \frac{M}{\pi R^2} \cdot R \cdot \frac{2\pi}{N} = \frac{2M}{RN} \\ F_{max} &= \frac{2(12\ 000)}{(8/2)(12)} = 500\ \text{lb} \quad \underline{\text{Ans.}} \end{aligned}$$

(c) $F = F_{max} \sin \theta$

$$\begin{aligned} M &= 2F_{max} R [(1) \sin^2 90^\circ + 2 \sin^2 60^\circ \\ &\quad + 2 \sin^2 30^\circ + (1) \sin^2 (0)] \\ &= 6F_{max} R \end{aligned}$$

from which

$$F_{max} = \frac{M}{6R} = \frac{12\ 000}{6(8/2)} = 500\ \text{lb} \quad \underline{\text{Ans.}}$$

The simple general equation resulted

in part (b)

$$F_{max} = \frac{2M}{RN}$$

B-34 THIS IS AN EXERCISE IN ESTIMATION FOR THE PURPOSE OF UNDERSTANDING THE CHANCE OF OVER-PROOFLOADING IS PRESENT

B-34

IS A FAST LOAD MAT WHE AND COM PROE THIN ON

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B-34 (CONT.)

IS A 4% CHANCE OF ONE OR MORE FASTENERS BEING OVER-PROOF LOADED. DESPITE THE APPROXIMATION, THIS IS A CAUTION THAT WHEN USING "MINIMUM" PROPERTIES AND LUBRICATED ASSEMBLY, AND COUNTING ON AVOIDING OVER-PROOF LOADING, IT IS A "SURE THING" THAT YOU CANNOT COUNT ON IT.

B-35 M20 ISO B.B COARSE THREADS, 3.4 MM WASHER UNDER BOLT AND NUT. JOIST FLANGES 16 MM THICK, HOUSING FLANGE 20 MM THICK. $E = 135 \text{ GPa}$.

(a) TO AVOID CONFUSION IN THIS PROBLEM WE WILL CALL PRELOAD P_i RESERVING F_i FOR THE CDF OF ORDERED FAILURE.

TABLE B-11: $S_p = 600 \text{ MPa}$

EQ. (B-41)

$$P_i = 0.9 A_t S_p = 0.9 (245) 600 (10^{-3}) = 132.3 \text{ kN}$$

$$T = 0.18 (132.3) 20 = 476 \text{ N}\cdot\text{m} \text{ Ans.}$$

(b) THIS OPENS AN INTERESTING PART OF STATISTICS CALLED ORDER STATISTICS, WHICH ADDRESSES ISSUES REGARDLESS OF THE DISTRIBUTION IDENTITY. THE CDF OF THE SMALLEST AND LARGEST ORDERED PRELOAD IS GIVEN BY, MEDIAN

$$\bar{F}_1 = \frac{i-0.3}{n+0.4} = \frac{1-0.3}{4+0.4} = 0.1591$$

$$\bar{F}_4 = \frac{4-0.3}{4+0.4} = 0.8409$$

IF OUR PRELOAD IS NORMALLY-DISTRIBUTED, FROM TABLE E-10

$\bar{z}_1 = -0.9981$ AND $\bar{z}_4 = +0.9981$. THE STANDARD DEVIATION OF P_i IS

$$C_p \mu_p = 0.09 (132.3) = 11.907 \text{ kN}$$

$$\bar{P}_i = \mu_p + \bar{z}_i \sigma_p = 132.3 + (-0.9981) 11.907 = 120.4 \text{ kN}$$

$$\bar{P}_4 = \mu_p + \bar{z}_4 \sigma_p = 132.3 + 0.9981 (11.907) = 144.2 \text{ kN}$$

THIS MEANS THAT ON MANY OBSERVATIONS OF FIRST AND FOURTH PRELOADS HALF WILL EXCEED 120.4 kN (AND 144.2 kN) AND HALF WILL NOT. THIS DESCRIBES THE BEHAVIOR OF MEDIANS.

IT IS POSSIBLE TO DESCRIBE MEAN VALUES AS WELL.

$$\bar{F}_1 = \frac{i}{n+1} = \frac{1}{4+1} = 0.2$$

$$\bar{F}_4 = \frac{4}{4+1} = 0.8$$

FROM TABLE E-10, $\bar{z}_1 = -0.8415$ AND $\bar{z}_4 = +0.8415$

$$\bar{P}_1 = 132.3 + (-0.8415) (11.907) = 122.3 \text{ kN}$$

$$\bar{P}_4 = 132.3 + 0.8415 (11.907) = 142.3 \text{ kN}$$

THIS MEANS THAT ON MANY OBSERVATIONS OF THE FIRST FAILURE, THE MEAN VALUE WOULD APPROACH 122.3 kN.

B-36 (a) ISO M20 X 2.5 GRADE 8.8 COARSE PITCH BOLTS, LUBRICATED.

TABLE B-2, $A_t = 245 \text{ mm}^2$

TABLE B-11, $S_p = 600 \text{ MPa}$

$$A_d = \pi (20)^2 / 4 = 314.1 \text{ mm}^2$$

$$F_p = 245 (0.600) = 147 \text{ kN}$$

$$F_i = 0.90 F_p = 0.90 (147) = 132.3 \text{ kN}$$

$$T = 0.18 (132.3) 20 = 476 \text{ N}\cdot\text{m} \text{ Ans.}$$

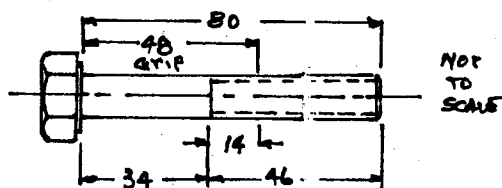
$$(b) L \geq L_g + H = 48 + 18 = 66 \text{ mm}$$

SET $L = 80 \text{ mm}$, TABLE E-17

$$L_T = 2D + 6 = 2(20) + 6 = 46 \text{ mm}$$

$$L_d = L - L_T = 80 - 46 = 34 \text{ mm}$$

$$L_T = L_g - L_d = 48 - 34 = 14 \text{ mm}$$



CHAPTER 9

9-1 EQ. (9-3):

$$F = 0.707 h \ell \tau = 0.707 (5/16) 4 (20) = 17.7 \text{ kip Ans.}$$

9-2 TABLE 9-7: $\tau_{all} = 21.0 \text{ kpsi}$

$$f = 14.85 h \text{ kip/in} = 14.85 (5/16) = 4.64 \text{ kip/in}$$

$$F = f \ell = 4.64 (4) = 18.56 \text{ kip Ans.}$$

9-3 TABLE E-20:

1018 HR; $S_{ut} = 58 \text{ kpsi}$, $S_y = 32 \text{ kpsi}$
 1018 CR; $S_{ut} = 64 \text{ kpsi}$, $S_y = 34 \text{ kpsi}$
 COLD-ROLLED PROPERTIES DEGRADE TO HOT-ROLLED PROPERTIES IN THE NEIGHBORHOOD OF THE WELD.

TABLE 9-5:

$$\tau_{all} = \min(0.30 S_{ut}, 0.40 S_y) = \min(0.30 [58], 0.40 [32]) = \min(17.4, 12.8) = 12.8 \text{ kpsi}$$

FOR BOTH MATERIALS.

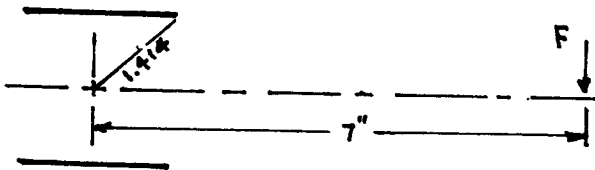
EQ (9-3): $F = 0.707 h \ell \tau_{all}$

$$F = 0.707 (5/16) 4 (12.8) = 11.3 \text{ kip Ans.}$$

9-4 EQ. (9-3)

$$\tau = \frac{\sqrt{2} F}{h \ell} = \frac{\sqrt{2} (32)}{(5/16) 2 (4)} = 18.1 \text{ kpsi Ans.}$$

9-5 $b = d = 2 \text{ in}$



(a) PRIMARY SHEAR

$$\tau' = \frac{V}{A} = \frac{F}{1.414 (5/16) 2} = 1.13 F \text{ kpsi}$$

SECONDARY SHEAR Table 9-2

$$J_u = \frac{d (3b^2 + d^2)}{6} = \frac{2 (3 \cdot 2^2 + 2^2)}{6}$$

$$= 0.53 \text{ in}^3$$

$$J = 0.707 h J_u = 0.707 (5/16) 5.3 = 1.18 \text{ in}^4$$

$$\tau_x'' = \tau_y'' = \frac{M r_y}{J} = \frac{7 F (1)}{1.18}$$

$$= 5.93 F \text{ kpsi}$$

MAXIMUM SHEAR

$$\tau_{max} = \sqrt{\tau_x''^2 + \tau_y''^2} = F \sqrt{5.93^2 + (1.13 + 5.93)^2} = 9.22 F \text{ kpsi}$$

$$F = \frac{\tau_{all}}{9.22} = \frac{20}{9.22} = 2.17 \text{ kip Ans.}$$

(b) E 7010, Table 9-3, $\tau_{all} = 21 \text{ kpsi}$ Table E-20:

MEMBER 1; $S_{ut} = 55 \text{ kpsi}$, $S_y = 30 \text{ kpsi}$
 MEMBER 2; $S_{ut} = 50 \text{ kpsi}$, $S_y = 27.5 \text{ kpsi}$
 E 7010; $S_{ut} = 70 \text{ kpsi}$, $S_y = 57 \text{ kpsi}$
 MEMBER 2 CONTROLS

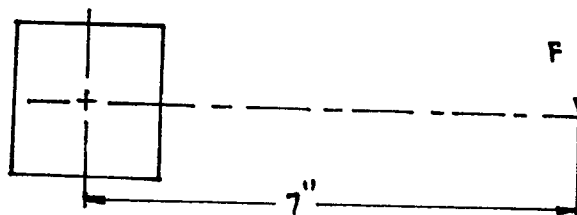
Table 9-5:

$$\tau_{all} = \min[0.30 (50), 0.40 (27.5)] = \min[15, 11] = 11 \text{ kpsi}$$

ALLOWABLE LOAD

$$F = \frac{\tau_{all}}{9.22} = \frac{11}{9.22} = 1.19 \text{ kip Ans.}$$

9-6 $b = d = 2 \text{ in}$



PRIMARY SHEAR

$$\tau' = \frac{V}{A} = \frac{F}{4 (0.707) 5/16 (2)} = 0.566 F$$

9-6 (CONT.)

SECONDARY SHEAR TABLE 9-2:

$$J_u = \frac{(b+d)^3}{6} = \frac{(2+2)^3}{6} = 10.67 \text{ in}^3$$

$$J = 0.707h J_u = 0.707(5/16) 10.67 = 2.36 \text{ in}^4$$

$$\tau_x'' = \tau_y'' = \frac{M r_y}{J} = \frac{(7F)(1)}{2.36} = 2.97F$$

MAXIMUM SHEAR

$$\begin{aligned} \tau_{max} &= \sqrt{\tau_x''^2 + \tau_y''^2} \\ &= F \sqrt{2.97^2 + (0.556 + 2.97)^2} \\ &= 4.62 F \text{ kpsi} \end{aligned}$$

$$F = \frac{\tau_{all}}{4.62}$$

WHICH IS TWICE $\tau_{max}/9.22$ OF Prob. 9-5.

9-7 ALTERNATING LOAD FATIGUE.

THROAT AREA OF WELDMENT

$$A = 0.707(6)(60+50+60) = 721 \text{ mm}^2$$

MEMBERS ENDURANCE LIMIT

$$\begin{aligned} S_{ut} &= 320 \text{ MPa}, S'_e = 0.506(320) = 161.9 \text{ MPa} \\ k_a &= 271(320)^{-0.995} = 0.872 \text{ MPa} \end{aligned}$$

$$k_b = 1 \text{ (DIRECT SHEAR)}$$

$$k_c = 0.256(320)^{0.125} = 0.531$$

$$k_d = 1$$

$$k_e = \frac{1}{k_{ts}} = \frac{1}{2.7} = 0.370$$

$$S'_e = 0.872(1)0.531(0.37)161.9 = 27.7 \text{ MPa}$$

ELECTRODE: 6010

$$S_{ut} = 62(6.89) = 427 \text{ MPa}$$

$$S'_e = 0.506(427) = 216 \text{ MPa}$$

$$k_a = 271(427)^{-0.995} = 0.654$$

$$k_b = 1 \text{ (DIRECT SHEAR)}$$

$$k_c = 0.258(427)^{0.125} = 0.550$$

$$k_d = 1$$

$$k_e = 1/k_{ts} = 1/2.7 = 0.370$$

$$S'_e = 0.654(1)0.55(0.37)216 = 28.7 \text{ MPa}$$

MEMBERS CONTROL. FOR A LOS OF 1

$$\begin{aligned} F_a &= \tau_a A = 27.7(721)10^{-3} = 19.9 \text{ kN} \\ &= 4.45 \text{ kip} = F_{max} \text{ Ans.} \end{aligned}$$

9-8 FROM PROBLEM 9-7

$$A = \frac{721 \text{ mm}^2}{645} = 1.18 \text{ in}^2$$

$$L = (60+50+60) = 170 \text{ mm} = 6.69 \text{ in}$$

$$F_{max} = 19.9 \text{ kN} = 4.45 \text{ kip}$$

FROM TABLE 9-8, CONFIGURATION 25, CATEGORY F, AND $K = -1$

$$\begin{aligned} (\tau_{max})_{all} &= \frac{\tau_{sr}}{1-K} = \frac{S}{1-(-1)} = 4 \text{ kpsi} \\ &= 27.6 \text{ MPa} \end{aligned}$$

SINCE $(\tau_{max})_{all} = \tau_{max}$, I.E. $27.6 \approx 27.7 \text{ MPa}$ THE WELD IS BARELY SATISFACTORY. THE WELDING CODE IS COMING AT THIS PROBLEM FROM A DIFFERENT PERSPECTIVE. IN THIS CASE THE RESULTS ARE COMPARABLE. IF WELDING CODE PREVAILS THE ALLOWABLE MAXIMUM LOAD F_{max} IS

$$F_{max} = 4(1.18) = 4.72 \text{ kip Ans.}$$

9-9 $\gamma' = 0$ (WHY)

$$\text{TABLE 9-2: } J_u = 2\pi r^3 = 2\pi(4)^3 = 4.02 \text{ cm}^3$$

$$J = 0.707h J_u = 0.707(0.5)4.02 = 1.42 \text{ cm}^4$$

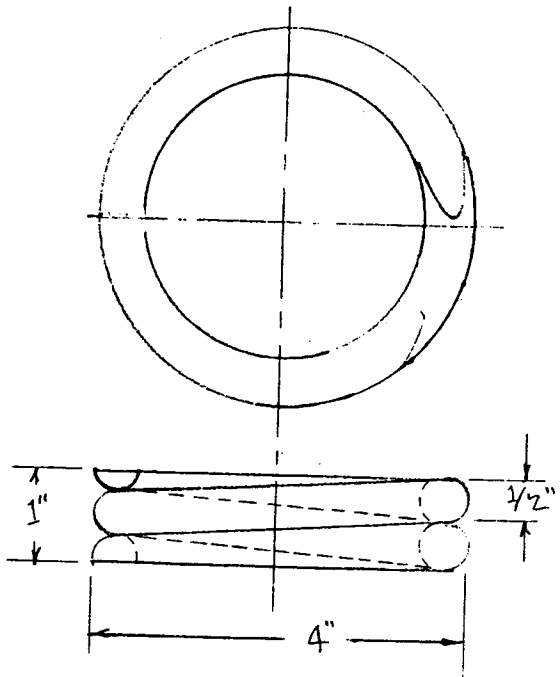
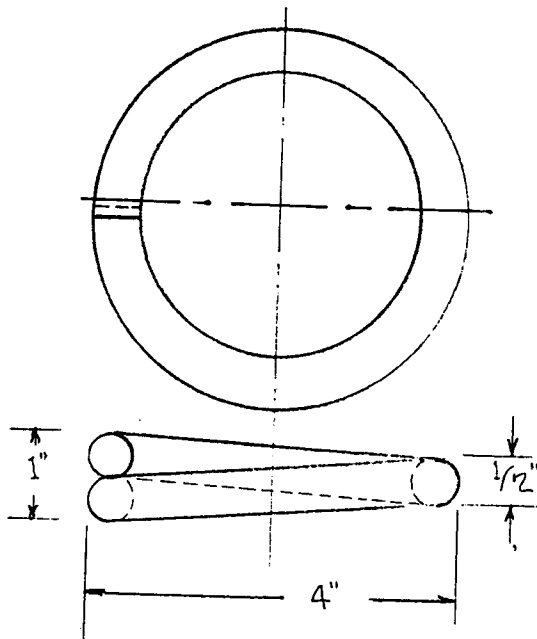
$$M = 200 \text{ F N}\cdot\text{m (FIN KN)}$$

$$\gamma'' = \frac{M r}{J} = \frac{(200 \text{ F})4}{2(1.42)} = 2.82 \text{ F}$$

$$F = \frac{\tau_{all}}{\gamma''} = \frac{140}{2.82} = 49.2 \text{ kN Ans.}$$

CHAPTER 10

10-1



$$10-2 \quad A = S d^m$$

$$\dim(A) = \dim(S) \dim(d^m) = \text{ksi} \cdot \text{in}^m$$

$$\dim(A_1) = \dim(S_1) \dim(d_1^m) = \text{MPa} \cdot \text{mm}^m$$

$$A_1 = \frac{\text{MPa}}{\text{ksi}} \cdot \frac{\text{mm}^m}{\text{in}^m} A$$

$$= 6.894757 (25.4)^m A$$

$$\approx 6.89 (25.4)^m A$$

FOR MUSIC WIRE, TABLE 10-5:

$A = 201$, $m = 0.145$; WHAT IS A_1 ?

$$A_1 = 6.89 (25.4)^{0.145} 201 = 2214 \text{ MPa} \cdot \text{mm}^m$$

ACCURACY DEPENDS OF VALUES OF A AND m FOR $AU \approx -9967$

REGRESSION ON 34 MUSIC WIRE SIZES, WHICH IS TABULATED AS 201 ksi $\cdot \text{in}^m$ AND $m = 0.144753 \approx 0.145$. THE REGRESSION VALUE OF $A = \text{EXP}(5.302493) = 200.8248$. THE IMPORTANT THING TO UNDERSTAND, IS THAT YOU CANNOT MULTIPLY 6.89 TIMES 201 TO GET A_1 .

10-3 GIVEN: MUSIC WIRE, $d = 0.105 \text{ in}$, $OD = 1.225 \text{ in}$, PLAIN GROUND ENDS, $N_t = 12$ coils. Page 596 TABLE: $Q = 1$, $Q' = 1$

$$N_a = N_t - Q = 12 - 1 = 11$$

$$L_s = (N_a + Q') d = (11 + 1) 0.105 = 1.26 \text{ in}$$

(a) TABLE 10-5: $A = 201$, $m = 0.145$

EQ. (10-17)

$$S_{ut} = \frac{201}{0.105^{0.145}} = 278.4 \text{ ksi}$$

EQ. (10-28)

$$S_{sy} = 0.45 (278.7) = 125.4 \text{ ksi}$$

$$D = 1.225 - 0.105 = 1.120 \text{ in}$$

$$C = \frac{D}{d} = \frac{1.120}{0.105} = 10.6$$

EQ. (10-6):

$$K_B = \frac{4(10.6) + 2}{4(10.6) - 3} = 1.126$$

$$(F)_{S_{sy}} = \frac{\pi d^3 S_{sy}}{8 K_B D} = \frac{\pi (0.105)^3 125.4 (10^3)}{8 (1.126) 1.120}$$

$$= 45.2 \text{ lb}$$

10-3 (CONT.)

$$k = \frac{d^4 G}{8 D^3 N_a} = \frac{0.105^4 11.75(10^6)}{8(1.120)^3 11}$$

$$= 11.55 \text{ lb/in}$$

$$L_o = \frac{(F)_{sy}}{k} + L_s = \frac{45.2}{11.55} + 1.26$$

$$= 5.17 \text{ in Ans.}$$

$$(b) (F)_{sy} = 45.2 \text{ Ans.}$$

$$(c) k = 11.55 \text{ lb/in Ans.}$$

$$(d) (L_o)_{cr} = \frac{2.63 D}{\alpha} = \frac{2.63(1.120)}{0.5}$$

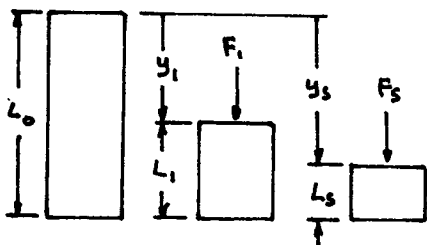
$$= 5.89 \text{ in}$$

MANY DESIGNERS PROVIDE $(L_o)_{cr}/L_o \geq 5$ OR MORE. PLAIN-ENDED SPRINGS ARE NOT OFTEN USED IN MACHINERY DUE TO BUCKLING UNCERTAINTY.

10-4 SEE PROB. 10-3. STATIC SERVICE SPRING. EQS. (10-24) TO (10-31) ONLY AS BASIS FOR ADEQUACY ASSESSMENT.

$$4 \leq C \leq 12: C = 10.6 \text{ O.K.}$$

$$3 \leq N_a \leq 15: N_a = 11, \text{ O.K.}$$



$$L_o = 5.17 \text{ in}, L_s = 1.26 \text{ in}$$

$$y_1 = \frac{F_1}{k} = \frac{30}{11.55} = 2.60 \text{ in}$$

$$L_1 = L_o - y_1 = 5.17 - 2.60 = 2.57 \text{ in}$$

$$\xi = \frac{y_s}{y_1} - 1$$

$$= \frac{5.17 - 1.26}{2.60} - 1 = 0.50$$

$$\bullet \xi \geq 0.15: \xi = 0.50, \text{ O.K.}$$

P. 592

$$\tau_1 = K_B \frac{8 F_1 D}{\pi d^3} = 1.126 \frac{8(30)(1.120)}{\pi 0.105^3}$$

$$= 83224 \text{ psi}$$

$$n_s = \frac{S_{sy}}{\tau_1} = \frac{125.4(10^3)}{83224} = 1.51$$

$$\bullet n_s \geq 1.2: n_s = 1.51 \text{ O.K.}$$

$$\tau_s = \tau_1 \frac{45.2}{30} = 83224 \frac{45.2}{30}$$

$$= 125391 \text{ psi}$$

$$S_{sy}/\tau_s = 125.4(10^3)/125391 \approx 1$$

$$\bullet S_{sy}/\tau_s \geq (n_s)_d:$$

NOT SOLID-SAFE. NOT O.K.

$$\bullet L_o \leq (L_o)_{cr}: 5.17 \leq 5.89$$

MARGIN COULD BE HIGHER, NOT O.K. DESIGN UNSATISFACTORY Ans. OPERATE OVER A ROD?

10-5 STATIC SERVICE SPRING, HD STEEL WIRE, $d = 2 \text{ mm}$, $OD = 22 \text{ mm}$, $N_t = 8.5$ turns, PLAIN AND GROUND ENDS, $\Phi = 1$, $\Phi' = 1$

PRELIMINARIES

$$\text{TABLE 10-5: } A = 1783 \text{ MPa} \cdot \text{mm}^m$$

$$m = 0.190$$

$$S_{ut} = \frac{1783}{2^{0.190}} = 1563 \text{ MPa}$$

$$S_{sy} = 0.45(1563) = 703.4 \text{ MPa}$$

$$D = OD - d = 22 - 2 = 20 \text{ mm}$$

$$C = 20/2 = 10$$

$$K_B = \frac{4C + 2}{4C - 3} = \frac{4(10) + 2}{4(10) - 3} = 1.135$$