

# USABILITY EVALUATION OF A FLIGHT-DECK AIRFLOW HAZARD VISUALIZATION SYSTEM

*Cecilia R. Aragon, University of California, Berkeley, California  
and NASA Ames Research Center, Moffett Field, California*

## Abstract

Many aircraft accidents each year are caused by encounters with unseen airflow hazards near the ground, such as vortices, downdrafts, low level wind shear, microbursts, or turbulence from surrounding vegetation or structures near the landing site. These hazards can be dangerous even to airliners; there have been hundreds of fatalities in the United States in the last two decades attributable to airliner encounters with microbursts and low level wind shear alone. However, helicopters are especially vulnerable to airflow hazards because they often have to operate in confined spaces and under operationally stressful conditions (such as emergency search and rescue, military or shipboard operations).

Providing helicopter pilots with an augmented-reality display visualizing local airflow hazards may be of significant benefit. However, the form such a visualization might take, and whether it does indeed provide a benefit, had not been studied before our experiment.

We recruited experienced military and civilian helicopter pilots for a preliminary usability study to evaluate a prototype augmented-reality visualization system. The study had two goals: first, to assess the efficacy of presenting airflow data in flight; and second, to obtain expert feedback on sample presentations of hazard indicators to refine our design choices.

The study addressed the optimal way to provide critical safety information to the pilot, what level of detail to provide, whether to display specific aerodynamic causes or potential effects only, and how to safely and effectively shift the locus of attention during a high-workload task. Three-dimensional visual cues, with varying shape, color, transparency, texture, depth cueing, and use of motion, depicting regions of hazardous airflow, were developed and presented to the pilots.

The study results indicated that such a visualization system could be of significant value in improving safety during critical takeoff and landing operations, and also gave clear indications of the best design choices in producing the hazard visual cues.

## Introduction

Turbulence and other wind-related conditions were implicated in 2,098 out of 21,380 aircraft accidents in the NTSB accident database from 1989-99 [1]. Addressing the controllability problems created by airflow disturbances has thus been a major issue in aviation safety. Disturbances in airflow, including weather-related hazards (e.g. thunderstorms, low level wind shear or microbursts), and locally-generated airwake hazards (such as downdrafts, hot exhaust plumes, wake vortices from other aircraft, turbulence and vortices from surrounding vegetation or structures near the landing site), have all been documented to be hazardous to aircraft of all categories and classes. The weather-related airflow disturbances have been relatively well studied and there is commercial hardware for detection of adverse weather or microbursts. Our research focuses on the smaller-scale airwake hazards, of which there has been less study (although they are more common), and specifically on the display of such airflow hazards to the pilot during approach to landing.

Airflow hazards are hard to detect simply because air is invisible. Its flow pattern is undetectable by pilots on a landing approach unless the air happens to pick up dust, smoke or other aerosols that are visible to the human eye. Being thus unable to directly see a factor of potentially great importance to them, pilots learn to use their intuition concerning airflow over obstacles near the takeoff or landing site, and they learn to pick up visual cues from the surrounding area. These methods are inadequate, however, as airflow-related accidents still occur.

## ***Hazard detection architecture***

A complete onboard airflow hazard detection system would consist of three major components: sensors; classification and analysis; and display (human interface). Our research addresses the display stage, but we describe the others here to illustrate the problem in context.

**Sensors/detection.** Recent technological advances in sensor technology, especially Doppler lidar [2], PIV (Particle Image Velocimetry) [3], and forward-looking microwave radar, offer the potential for aircraft-based sensors which can gather large amounts of airflow data in real-time. There are currently available commercial systems utilizing this technology to detect moderate-scale airflow disturbances such as microbursts and windshear [4]. Current research into airflow detection techniques such as lidar is promising; it is believed that within a few years hardware capable of being mounted on an aircraft will be able to reliably scan the area a few hundred feet ahead of the aircraft and sample air particle vector velocities at one-foot intervals or less [5]. With the development of such devices, onboard detection systems that can convey detailed, specific information about airflow hazards to pilots in real-time become a possibility.

**Classification/analysis.** This area concerns the development of algorithms to input the particle positions and vector velocities, other variables such as density altitude, aircraft gross weight, and power available; to compute the locations of the areas of flow which may produce a hazard to this particular aircraft on this particular day; and to output the three-dimensional coordinates of the hazard location in real-time.

**Display to the pilot (user interface).** Given the airflow data and the known hazard areas, the problem then becomes to organize this vast amount of data, describing millions of particles swirling in different directions, and present it to the pilot in a manner that does not interfere with the primary task of operating the aircraft safely. An interface is required that can present potentially large amounts of data to the pilot in a non-intrusive yet comprehensive manner in real-time.

## ***Motivation for visual interface usability study***

Since an airflow hazard detection system generates a large amount of disparate data that must be organized and presented to a human operating a complex machine in a high-workload environment, an efficient method of human-machine communication is required. The human visual system has the highest bandwidth of all the senses. It can process gigabytes of data in real-time and organize it into patterns that the brain can use to draw conclusions and act very quickly. Beyond this general observation about the efficacy of visual input, we also note that the operation of landing an aircraft is an essentially visual operation, even if the flight itself is made under instrument weather conditions; the pilot looks at least at the instruments, and finally always at the landing site. It therefore makes sense to organize the airflow data into some type of visual display.

As with any type of user interface, usability evaluation is important to ensure that the display most efficiently supports the human operator's performance. Given the demanding environment and the relatively small population of highly trained pilots, it is especially critical to conduct a usability study before designing an airflow hazard display system.

## ***Shipboard rotorcraft operations***

Although the need to detect airflow hazards exists for all pilots in all aircraft, for our research we chose to focus on helicopter operations, and specifically on Navy shipboard rotorcraft operations, to which the Navy refers as the "Dynamic Interface." There were several reasons for this choice.

Helicopters are especially vulnerable to airflow disturbances; first, by the nature of the aerodynamic forces involved, and second, because helicopters are often called upon to operate into and out of confined areas or areas that naturally have disturbed airflow. For example, emergency search and rescue may have to operate in mountainous areas and small clearings surrounded by vegetation and cliffs where the winds are always high. Helicopters also must land on urban rooftops, offshore oil platforms, or on the decks of ships. A device for detecting airflow

hazards therefore has a special utility for helicopter operations.

Operating a helicopter off a moving aircraft carrier is one of the most demanding tasks a helicopter pilot can face [6]. Because the ship is moving, its superstructure will always generate an airwake consisting of vortices and other unseen hazards. In addition, high sea states may cause extreme ship motion, and low visibility may degrade visual cues. The pilot must maneuver the helicopter within very tight tolerances under adverse conditions. It is a task that demands the utmost concentration and skill from the pilot [Figure 1]. A system that can deliver even an incremental amount of assistance to the pilot in this high-demand environment could have a significant impact on safety.



**Figure 1. Shipboard helicopter landing**

Helicopter accidents and incidents that occur each year range from fatal accidents to incidents such as "tunnel strikes" (when a rotor blade strikes the fuselage of the helicopter). There have been approximately 120 tunnel strikes since 1960, causing damage ranging from \$50-\$75K to over \$1M [5]. When analysis of these accidents and incidents is performed, the conclusion is frequently that they were due to unseen airflow hazards such

as vortices, downdrafts, hot exhaust plumes, or wind shear, where the pilot and ground crew were initially unaware of the danger and the pilot was unable to react in time. Presenting the appropriate information to the pilot or flight deck air boss in advance of the hazard encounter could reduce or prevent these types of accidents in the future.

Finally, because shipboard rotorcraft operations are such a demanding environment, the area is very well studied. The Navy has compiled significant amounts of data from shipboard flight tests, wind tunnel tests, and computational fluid dynamics computations studying the airflow around moving ships of all types, and how the airwake changes when helicopters of different makes and models land on the ships. Utilizing the data from these extensive tests and computational studies, the Navy develops operational envelopes listing allowable wind conditions for many ship-rotorcraft combinations [7]. However, the envelopes are of necessity (for safety reasons) relatively narrow, and convey fairly limited information, basically a go/no-go decision. The envelopes do not state which safety considerations caused a particular operational limit, thus limiting the information available to the pilot. On the other hand, accidents and incidents occur during operations *within* the envelope every year. On occasion, during the post-accident analysis, the flight test engineers can point to existing airwake data to show that the accident was caused by disturbed airflow over a portion of the deck. In other words, the information that could have prevented the accident was known, but it had not been communicated to the pilot. Thus as Navy flight test engineers seek ways to increase fleet safety, this problem is ripe for solution.

In this paper, we begin by discussing previous research relevant to developing a visual hazard display to solve the airflow hazard problem. Then we describe our experimental procedure, discuss the results obtained, and finally give conclusions and directions for further work.

## **Previous Research**

There are three major areas where there are significant bodies of research that are relevant to our current efforts: flow visualization, aviation displays, and Navy shipboard rotorcraft operations.

## ***Flow visualization***

There is a large body of research on flow visualization, which often consists of detailed imagery of two- and three-dimensional airflow patterns, both static and dynamic, steady and unsteady, all designed to help scientists or engineers understand—and analyze at length—a particular instance of a fluid flow. Examples include streamlines and contour lines for the case of instantaneous flow [8], [9] and streaklines, timelines [10], flow volumes [11], spot noise [12], or moving textures for unsteady flow. The imagery is often quite complex and not suitable for rapid glances during time-critical tasks. Other potentially applicable work includes automated detection of swirling flow [13] and terrain and turbulence visualization [14].

Not being designed for viewing in the cockpit or under time pressure, flow visualization methods emphasize the informational richness of the imagery rather than the user interface. These methods have seldom if ever been subjected to usability studies, and are not in any case expected to be viewed and acted upon in real-time.

## ***Aviation displays***

It has long been recognized that applying developing technology to improve aviation displays might enhance aviation safety. Significant work in this area includes synthetic vision and augmented-reality displays (terrain in low-visibility environments, navigation aids) [15], [16], [17], [18], weather visualization including NASA's AWIN, TPAWS and AWE [19], [20], [21], [22], and turbulence detection/prediction [23]. Holforty's research on wake vortex visualization [24] is one of the first to provide three-dimensional hazard cues with the intent of eventually layering the cues on an augmented-reality display. There is also a large body of relevant work concerning human factors in the cockpit, including the study of attention and cockpit visual displays [25], [26], [27], [28].

Also, it has been known since the eighties that weather-related airflow phenomena such as microbursts and wind shear have been responsible for the loss of hundreds of lives in airliner accidents [29]. As a result, a great deal of work has been done to detect, predict, and display this type of

information to the pilot [30]. There are commercially available aircraft-based, forward-looking microwave radar and lidar systems that can detect microbursts and windshear. However, the emphasis was placed on integrating the information into existing cockpit displays, in order to reduce time to commercial deployment. Accordingly, no usability studies were focused specifically on the display itself or on whether a three-dimensional head-up display would be helpful in presenting hazard information to the pilot.

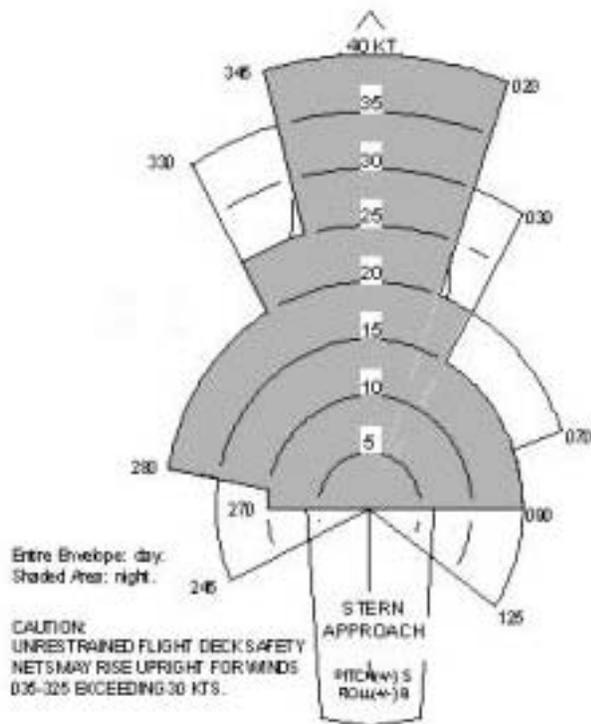
## ***Navy shipboard rotorcraft operations***

Because landing a helicopter on a moving ship deck is a very hazardous environment [6], the Navy has long operated a program to perform flight testing of these conditions [7]. They have run a significant amount of flight testing with the stated goal of improving flight safety. The ship superstructures always produce airwake. In addition, aircraft landing on shipboard are plagued by hot exhaust plumes, very powerful shipboard radar that interferes with aircraft systems, inaccurate anemometers, and of course the inevitable problems associated with high sea states such as extreme values of ship pitch, heave, and roll, and high, turbulent winds, as well as the requirement to land in a very confined area.

Because understanding the airwake over the ship is so critical, the Navy uses techniques including shipboard flight testing, wind tunnel tests, computational fluid dynamics, and sampling of the airflow vector velocities at various points in the flow field behind the superstructure in the helicopter landing zones with handheld anemometers. Lidar detectors are being developed and improved and research is ongoing in this area.

The current method of communicating this information to the pilots consists of operational envelopes for each ship-rotorcraft combination [Figure 2].

The envelope depicts the allowable wind speeds and directions that a given helicopter is allowed to land on a particular ship. It is necessarily conservative, as the envelope has to include all flight conditions and all fleet pilot skill levels. The envelopes limit allowable landing conditions significantly; however, even with this



**Figure 2. Ship-rotorcraft operational envelope**

cautious approach, accidents due to airflow hazards still occur. In one recent example, a helicopter was damaged when starting up on shipboard, even though the winds were within the allowable starting envelope. Another helicopter was operating upwind of the first, and this configuration caused hazardous airflow to be present at the downwind spot [5]. There was knowledge of this problem from the Navy flight testing program; however, the envelopes cannot portray every combination of aircraft location on a ship that may have many helicopters and aircraft in operation at the same time.

## Experiment Design

During potentially hazardous conditions, high winds, low visibility, or extreme ship motion, the pilot's attention will naturally be focused outside during the critical landing moments; he or she will not want to look down at a cockpit instrument display. In designing our experiment, we assumed pilots would prefer an augmented-reality hazard visualization display (as was verified during the usability study). However, the head-up display must be carefully designed not to distract from the

key shipboard visual cues, which may be degraded during a challenging nighttime or poor-weather landing on a ship. Studies have shown that head-up displays with superimposed symbology may on occasion cause performance problems due to attentional capture by the perceptual grouping of the superimposed symbols [31]. "Scene-linked" head-up displays, or displays where there is no differential motion between the superimposed symbology and the outside scene, can avoid this type of distraction. For this reason we decided to develop a head-up display where the hazard indicator is three-dimensional and appears to be physically part of the world.

## Rapid Prototype Phase

We first constructed a horizontal prototype (a relatively full-featured simulation of the interface with no underlying functionality) [32] of an augmented-reality hazard visualization system that included many different types of hazard indicators. The usability study on the horizontal prototype had two main goals: first, to determine whether presenting airflow hazard data to helicopter pilots would be helpful to them; and second, to obtain expert feedback on the presentation of sample hazard indicators, from which we could refine our design choices.

We decided to perform interactive prototyping [32], a technique where the prototype is altered on the fly as the test user comments on its effectiveness. This enabled us to rapidly modify the design and obtain feedback on multiple variations in a single session.

## Selection of Platform for Rapid Prototype

The next task was to identify the best tool for creating a relatively realistic, three-dimensional visual simulation of the helicopter pilot's view out the cockpit windscreen during the final approach to a shipboard landing. The tool was required to support rapid prototyping, 3D modeling, and simple animation. It was especially important that we be able to create new hazard visualizations within minutes, as we were hoping to get feedback from the study participants and implement their suggestions during the session so as to tighten the feedback loop.

Consultation with Navy flight test engineers provided detailed descriptions of what a landing approach should look like. Additionally, we were provided with an extremely detailed 3D CAD model of a Spruance-class destroyer (DD 963). An ideal prototype platform would be able to use this data to render a realistic approach.

Three approaches were considered for the prototype software platform: a low cost, off-the-shelf flight simulator; a 3D animation system; and a 3D CAD tool.

The Microsoft Flight Simulator was considered because it offered the possibility of the pilots being able to use a joystick, and potentially the opportunity to alter the visual hazard display without affecting the flight simulation. However, there was no convenient interface for importing the existing ship model into MS Flight Simulator.

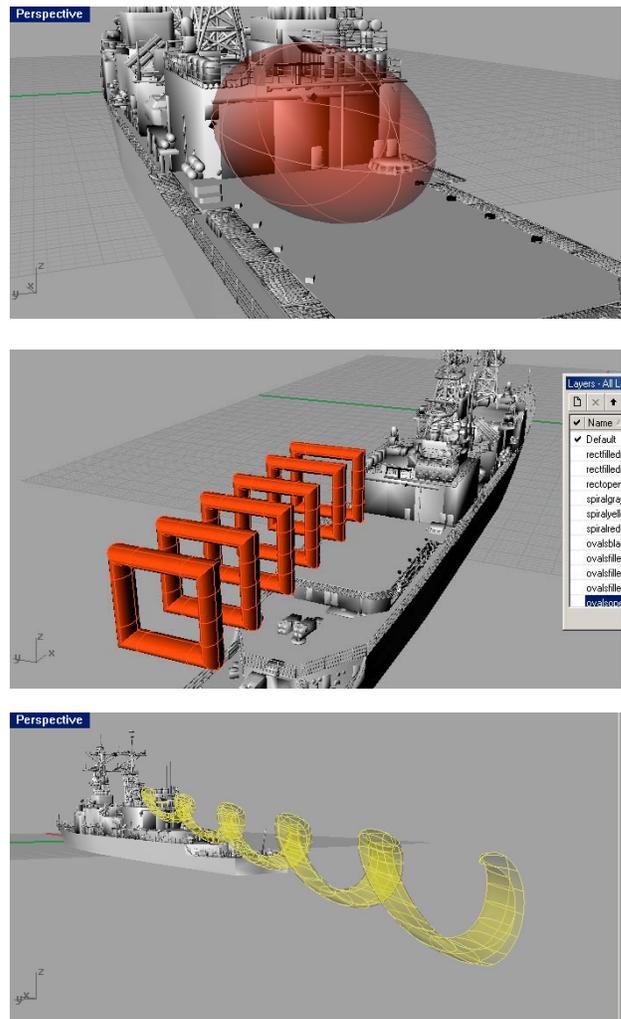
We also investigated various 3D animation systems such as WildTangent, VRML, and Flash. However, although these systems could handle the animation well, the overhead for changing hazard indicators was considerable, essentially comparable to working in a programming language. (Actual programming languages, such as Java, were ruled out for the same reason.)

The CAD modeling tool we selected, Rhino3D, offered rapid construction and alteration of the prototype scenarios and easy access to the ship model data. It was very easy and quick to create many different types of hazard indicators and modify their shape, location, color, texture, and transparency. Although not a flight simulator, the CAD program allowed us to simulate the final approach to landing by rotating and zooming the model of the ship with the hazard indicator displayed above it.

### **Methodology**

We recruited three highly experienced (>1700 hours) helicopter pilots and flight test engineers, all with shipboard landing experience. Each session with a participant pilot consisted of a 1 ½-hour interview with the pilot in front of a projection screen. All sessions were videotaped. Two experimenters conducted the session, one operating the computer and the other interviewing the pilot and taking notes.

The operator-experimenter used the Rhino3D CAD program to display on the projection screen a model of the ship (DD 963), with a hazard indicator displayed on the ship's deck where hazardous airwake might be found. The operator manually simulated a helicopter's view of an approach to landing on shipboard as the pilot watched and commented. A wide selection of different types of hazard indicators were stored in layers in Rhino3D, so that features could be selectively turned on and off by the operator. The features that were varied in the hazard indicator included shape, location, color, texture, transparency, depth cueing, and motion.



**Figure 3. Hazard indicators in Rhino3D**

Feedback was solicited from the pilot. If the pilot suggested a change, the operator implemented it on the fly and the pilot was asked to judge whether the change was an improvement. The experimenter asked both specific and open-ended

questions throughout the interview designed to elicit the pilots' expertise.

Using the pilots' responses, we attempted to assess the efficacy of presenting airflow data in flight, and to select the most efficacious (as judged by the pilots) visual presentation for the hazard indicator.

### ***Participants***

In choosing participants, we sought pilots with a great deal of helicopter experience and, ideally, experience with shipboard landings of large military helicopters. Finding pilots with the requisite domain-specific knowledge was challenging. The final test group for the prototype consisted of two military pilots and one experienced civilian helicopter pilot:

**Participant 1:** Navy helicopter test pilot, 2000 hours of flight time, 17 years experience.

**Participant 2:** Navy helicopter flight test engineer, 4000 hours of helicopter simulator time, 100 hours of flight time, 17 years experience with shipboard helicopter flight tests.

**Participant 3:** Civilian helicopter flight instructor, 1740 hours of flight time, 3 years experience.

### **Results**

Results of the usability study on the rapid prototype were encouraging, but in some respects surprising as to the types of display features pilots found helpful. All participants said they liked the system and would use it if it were installed on their aircraft. As they viewed the interface, the pilots repeatedly stated that they wanted such a hazard visualization tool.

As to the type of visualization, the strongest overriding principle that emerged from this experiment is that helicopter pilots are using all their attention to focus on the extremely demanding task of landing on a moving ship deck, perhaps under low visibility conditions or at night, and the hazard indicator must not distract from that focus. To that end, the participants favored much simpler imagery than we would have expected.

The pilots strongly rejected the use of flow visualization indicators, and especially of motion to indicate flow. Given the manner in which fixed-wing pilots look for natural flow indicators such as dust devils near the runway, smoke plumes, wind-blown vegetation etc., we had anticipated that helicopter pilots would prefer a dynamic flow visualization, capable of indicating the direction and velocity of particles in the hazardous region. However, the participants exhibited resistance to such a design. All participants, even while reiterating their desire for 3D hazard visualization, stated that motion was distracting during the approach and particularly during the critical moments near touchdown. A static visualization, even supplying less information, was strongly preferred over a dynamic hazard indicator. That is, the participants sought a real-time decision support tool, not an airflow analysis tool.

Below we describe the hazard display parameters that were varied in the prototype usability test, and the results obtained for each.

### ***Color***

We showed hazard indicators in single and in multiple hues, using colors spanning the spectrum. All pilots preferred single-color hazard indicators, and indeed, preferred only two colors for the final system: yellow for caution and red for danger. Yellow, according to the participants, should indicate an airflow hazard that could necessitate strong pilot input to stay safe, but where the aircraft should maintain controllability. Red should indicate danger, an airflow hazard that would likely be beyond the limits of the aircraft and would put its controllability in question.

We were surprised to find the pilots unanimous on the point that a hazard indicator should be rendered in a single color (either red or yellow). Multiple-color hazard indicators were considered distracting and confusing. When the experimenters pointed out that a vortex core could have very strong winds but the outer portion of the vortex might not be as hazardous, so that a two-color indicator with a red core and a yellow mantle might be useful, the pilots all disagreed, saying the red vortex core would be difficult to see or to locate correctly in a three-dimensional object. In addition to the overall view that the display would be

confusing, a concern was also expressed that a two-color indicator could tempt a pilot to venture into the yellow mantle while attempting to skirt the red core. That is, the two-color indicator was thought to potentially support an incorrect decision to land in dangerous conditions.

### ***Transparency***

While holding other variables constant, we varied the transparency of the displayed hazard indicator from 20% to 80% (according to the Rhino software controls). This test was repeated for a range of objects. The pilots preferred an average transparency near 70%. While desiring a hazard indicator sufficiently opaque to come to the pilot's attention, participants noted the critical need for the pilot to be able to see visual cues on the ship behind the hazard indicator.

### ***Depth Cues***

We displayed hazard indicators that hovered above the deck and cast no shadow, and others that had a colored shadow projected onto the deck directly below the indicator. Of those with shadows, some had a connecting vertical line from the indicator to the deck shadow. All of the pilots preferred shadows below objects, stating that they helped the pilot to localize the 3D indicator in space. Pilot #1 said shadows alone might be sufficient for a shipboard hazard warning system: "just paint the deck red if I need to wave off." Pilot #2 liked the idea of a connecting line between the hazard indicator and the deck. No participant wanted tick marks or numeric information floating with the hazard indicators. Again, they preferred to keep it simple; the purpose is to let the pilot see the location and approximate severity of a hazard, not to help them measure or analyze it.

### ***Texture***

We displayed hazard indicators having a series of arrows textured onto their partially transparent surface, to indicate the direction of airflow in that hazardous area, and asked pilots to compare them to indicators without the texture. Pilots #1 and #2 did not want the extra detail, saying it could be confusing or distracting. Pilot #3 suggested striping as a possible symbology, reminiscent of the yellow

and black caution tape that is a common symbol to most Americans.

### ***Shape***

We asked the pilots to comment on the effect of varying the shape of the hazard indicator, such as rectilinear transparent boxes, cloud shapes with rounded corners, spirals, rings both round and rectangular. The rectilinear and cloud shapes were favored over all others. Again, a preference for simplicity was displayed. One of the pilots pointed out that the floating rings looked a little bit like the HUD symbology for the "highway in the sky," perhaps beckoning the pilot to fly into the rings, the exact opposite of the intended action! This comment made clear the need to research all HUD symbology so as to avoid conflicts with existing symbology or commonly accepted designs.

### ***Motion***

There was a strong consensus that motion in the display, particularly fast motion, was distracting. Pilot #1 (the participant with the most experience landing on shipboard in actual hazardous conditions) said the visual indicators should absolutely not use motion at all. It was distracting, and in the worst case could induce vertigo, especially at night or in low-visibility situations. The pilot stated that if the indicators had to change their position in real-time to indicate a change in the location of the hazard, they should move smoothly, and attention should be paid to the edges to make sure no flashing or other video artifacts appear that might distract the pilot from the task of landing. This pilot also stated that the indicators should fade in and out gradually in response to changing hazard conditions (unless the pilot turned them on or off.) A sudden appearance of a hazard indicator, where there had been none, could be startling and potentially dangerous. Likewise any rapid motion or disappearance out of the corner of the pilot's eye during the landing could be distracting and potentially dangerous. Pilot #2 concurred that there should be no motion in the hazard indicators. Pilot #3, the civilian pilot, stated that *slow* motion on the surface of the indicator could conceivably be helpful to give an indication of which way the airflow was moving within, but

that in general, fast motion could be distracting and dangerous.

### ***Audio***

Some existing hazard warning systems for commercial aircraft use audible warnings, e.g. a bell or voice. Participants in our study were asked whether they would judge an audio indicator to be helpful or distracting. The consensus was against using audio. Pilots #1 and #2 were clear that they did not want the hazard indicator to have any audio component. Pilot #3 conjectured that a limited audio, such as a soothing female voice, might be helpful under certain limited conditions.

### ***General considerations***

Other comments the pilots made were that the indicator should appear at the 180-degree point, the point in the approach where the pilot is abeam the intended landing spot facing downwind. The indicators should then either turn off as the wheels cross the deck, or remain on throughout the landing. For yellow (caution indicated, but controllable) conditions, it was thought potentially helpful to leave the hazard indicator on display, as the pilot might choose to fly into the indicated area (the "curtain"). Numeric indicators representing airflow speed were not preferred; the pilots stated that they wouldn't have the time to read numbers as they approached the landing spot. All of the pilots preferred an idealized representation rather than exact visualization of airflow, again in the interest of keeping the display simple. One pilot suggested just painting the deck or the landing spot red or yellow. It was also suggested that more detailed options might be useful at the start of the approach. Perhaps a helicopter silhouette on the deck, or wind arrows or airflow lines, could be selected by the pilot at that point, fading to a simpler version as the pilot flew closer. It was also pointed out that it was important for the system to be credible, with no false positives or negatives. Finally, it was critical that the pilot be able to turn the system on and off, and that a vernier control be present to adjust the brightness of the display based on the ambient light.

## **Conclusions and Future Work**

A preliminary usability study of an airflow hazard visualization system for helicopter pilots landing on board a moving ship indicated that pilots would use such a system if it were available on their aircraft. They expressed a need to know more about airwake hazards and a desire to have the information presented to them in the cockpit as they were landing. The preference was for a head-up display with "scene-linked" indicators vs. an instrument panel display.

The pilots indicated that any airflow hazard symbology should present the minimum critical information such as location of the hazard and whether it was a warning (yellow) or danger (red). There was no desire for detailed quantitative information or even qualitative information such as type of hazard such as vortex, downdraft, turbulence, wind shear, etc. In other words, what the pilots are looking for is a decision support system, not a scientific visualization system, and any future work in this area should be done with this kept in mind. They want to be shown the effects – e.g. hazards to aircraft – and not causes – e.g. this is a vortex caused by the wind curling up and over the deck edge with downdrafts of up to 400 ft/minute. Extensive detail, motion, complex shapes, too many colors, were all considered too distracting and possibly dangerous in the high-demand environment of shipboard helicopter operations. Preference was strongly given to static rather than dynamic indicators. Concerns were expressed over distractions such as motion inducing vertigo, confusing symbology causing doubt in the pilot's mind, etc. Nevertheless, there was a clear desire to have such a system in the cockpit.

### ***"The Holy Grail"***

We close with a quote from one of the pilots in our usability study, asked if he thought a system of airflow hazard visualization might have the potential to improve helicopter flight safety:

*"...[This system] offers ... a chance to avoid mishaps that have happened before, combined with the opportunity to provide a greatly expanded operating envelope..."*

*"That's the holy grail... to be able to both increase safety and increase operational capability"*

*at the same time. Usually you don't find something with the potential to do both. Usually you either have something that makes it a lot safer but tends to impose certain operational restrictions...or you have something that gives you greater operational capability but there's risks associated with employing that additional capability... In this case you actually have a concept that could potentially give you both."*

Further work is indicated and currently the author is conducting a flight simulation study using the preferred set of hazard indicators. Airflow data and ship and helicopter aerodynamic models have been loaded into a high-fidelity rotorcraft flight simulator [33] and scenarios have been created where airflow hazards are known to be present near the landing sites on shipboard. A visual hazard indicator system has been developed and implemented, and integrated into the display system of the simulator. Experienced helicopter pilots with shipboard landing experience have been recruited to fly multiple approaches to a moving aircraft carrier under extreme wind and turbulence conditions, both with and without visual hazard indicators. Data is being gathered both subjectively by having the pilots fill out questionnaires about the hazard visualization system, and objectively by measuring flight path deviations, control surface motion and pilot workload, landing dispersion, and vertical speed at touchdown.

## Acknowledgments

This work was funded by the NASA Ames Full-Time Graduate Research Program. The author wishes to thank her adviser at UC Berkeley, Prof. Marti Hearst, for guidance and suggestions. This work could not have taken place without the active support and advice of Kurtis R. Long of the Navy Dynamic Interface Flight Test Group and Fluid Mechanics Laboratory at NASA Ames who generously shared his extensive knowledge. Thanks are also due to Advanced Rotorcraft Technology, Inc. for the use of their high-fidelity helicopter flight simulator. The author deeply appreciates their interest in and support of her research.

## References

- [1] FAA National Aviation Safety Analysis Center, <https://www.nasdac.faa.gov/>.
- [2] NASA Marshall Space Flight Center Lidar Tutorial, [http://www.ghcc.msfc.nasa.gov/sparcle/sparcle\\_tutorial.html](http://www.ghcc.msfc.nasa.gov/sparcle/sparcle_tutorial.html).
- [3] Particle Image Velocimetry at NASA Glenn, <http://www.grc.nasa.gov/WWW/OptInstr/piv/backgroud.htm>.
- [4] Chambers, J., 2003, *Concept to Reality*, NASA SP-2003-4529, [http://oea.larc.nasa.gov/PAIS/Concept2Reality/win\\_d\\_shear.html](http://oea.larc.nasa.gov/PAIS/Concept2Reality/win_d_shear.html).
- [5] Long, Kurtis R., 2003, Navy flight test engineer, personal communication.
- [6] Wilkinson, C. H., S. J. Zan, N. E. Gilbert, J. D. Funk, 1998, "Modelling and Simulation of Ship Air Wakes for Helicopter Operations."
- [7] Williams, Suni L., Kurtis R. Long, 1997, "Dynamic Interface Flight Tests and the Pilot Rating Scale," *American Helicopter Society 53<sup>rd</sup> Annual Forum*, Virginia Beach, VA.
- [8] Buning, P., 1989, "Numerical algorithms in CFD post-processing," *Computer Graphics and Flow Visualization in Computational Fluid Dynamics*, von Karman Institute for Fluid Dynamics Lecture Series 1989-07.
- [9] Strid, T., A. Rizzi, J. Ooppelstrup, 1989, "Development and use of some flow visualization algorithms," *Computer Graphics and Flow Visualization in Computational Fluid Dynamics*, von Karman Institute for Fluid Dynamics Lecture Series 1989-07.
- [10] Lane, David A., 1996, "Visualizing Time-Varying Phenomena In Numerical Simulations Of Unsteady Flows," NASA NAS Technical Report NAS-96-001.
- [11] Max, N., B. Becker, R. Crawfis, 1993, "Flow Volumes for Interactive Vector Field Visualization," *Proceedings of Visualization '93*, IEEE Computer Society Press, Los Alamitos, CA, pp. 19-23.
- [12] Post, F., J. van Wijk, 1993, "Visual representation of vector fields: recent developments

and research directions,” *Scientific Visualization Advances and Challenges*, Academic Press, San Diego, pp. 367-390.

[13] Sujudi, D., R. Haimes, 1995, “Identification of Swirling Flow in 3D Vector Fields,” *AIAA Paper 95-1715*, San Diego, CA.

[14] Turner, B., M.Y. Leclerc, M. Gauthier, K. Moore, and D. Fitzjarrald, 1994, “Identification of Turbulence Structures above a Forest Canopy using the Wavelet Transform,” *J. Geophys. Res.* 99 D1, pp. 1919-1926.

[15] Uenking, M. D., M. F. Hughes, 2002, “The Efficacy of Using Synthetic Vision Terrain-Textured Images to Improve Pilot Situation Awareness,” SAE World Aviation Congress.

[16] Bailey, R.E., R. V. Parrish, L. J. Kramer, S. D. Harrah, J. J. Arthur III, 2003, “Technical Challenges In the Development of a NASA Synthetic Vision System Concept,” NATO RTA.

[17] Spitzer, C. R., R. V. Parrish, D. G. Baize, M.S. Lewis, 2001, The Avionics Handbook: Synthetic Vision, NASA Aviation Safety Program.

[18] Alter, K. W., A. K. Barrows, C. Jennings, J. D. Powell, August 2000, “3-D Perspective Primary Flight Displays for Aircraft,” *Proceedings of Human Factors and Ergonomics Society*, San Diego, CA.

[19] Latorella, Kara, Jim Chamberlain, 2002, “Graphical Weather Information System Evaluation: Usability, Perceived Utility, and Preferences from General Aviation Pilots,” NASA AWIN technical report.

[20] Britt, Charles L., Carol W. Kelly, 2002, “User's Guide For An Airborne Doppler Weather Radar Simulation (ADWRS),” NASA TPAWS.

[21] Shaw, C. D., F.T. Jiang, R. M. Parry, B. Plale, A. Wasilewski, W. Ribarsky, N. L. Faust, 2001, “Real-Time Weather Data on Terrain,” NASA AWIN, SPIE Vol. #4368, Radar Sensor Technology and Data Visualization.

[22] Spirkovska, L., S. K. Lodha, 2002. AWE: Aviation weather data visualization environment, *Computers and Graphics*, 26, pp. 169-191.

[23] Switzer, G., C. Britt, 1996, “Performance of the NASA Airborne Radar with the Windshear

Database for Forward-Looking Systems,” NASA CR 201607.

[24] Holforty, Wendy, 2003, “Flight-Deck Display of Neighboring Aircraft Wake Vortices,” Ph.D. dissertation, Stanford University.

[25] Roscoe, S. N., 1968, “Airborne displays for flight and navigation,” *Human Factors*, 10, pp. 321-322.

[26] Wickens, C. D., R. Carbonari, D. Merwin, E. Morphew, J. O'Brien, 1997, “Cockpit Displays to Support Hazard Awareness in Free Flight,” NASA Technical Report ARL-97-7/NASA-97-4.

[27] Endsley, Mica R., 2001, “Designing for Situation Awareness in Complex Systems,” *Proceedings of the Second International Workshop on symbiosis of humans, artifacts and environment*, Kyoto, Japan.

[28] Lee, Alfred T., 1991, “Aircrew Decision-Making Behavior in Hazardous Weather Avoidance,” *Aviation, Space, and Environmental Medicine*.

[29] United States Department of Commerce, 1986, The Crash of Delta Flight 191 at Dallas-Fort Worth International Airport on 2 August 1985: Multiscale Analysis of Weather Conditions, National Oceanic and Atmospheric Administration, Boulder, CO.

[30] Wallace, L., 1993, Airborne Trailblazer, NASA Langley Wind Shear Program, <http://oea.larc.nasa.gov/trailblazer/SP-4216/chapter5/ch5.html>.

[31] McCann, R. S., D. C. Foyle, 1994, “Superimposed symbology: Attentional problems and design solutions,” *SAE Transactions: Journal of Aerospace*, 103, 2009-2016.

[32] Nielsen, Jakob, 1993, Usability Engineering, Morgan Kaufmann, San Francisco, CA, pp. 93-98.

[33] Advanced Rotorcraft Technology, Inc., 1685 Plymouth St., Suite 250, Mountain View, CA 94043, <http://www.flightlab.com>.

## Keywords

Augmented reality, flight-deck displays, usability, human-computer interface, airflow hazards, data visualization, helicopters, human factors in aviation.

## **Author's Biography**

Cecilia Aragon is a computer scientist at NASA Ames Research Center and a Ph.D. candidate in the Computer Science Division at the University of California, Berkeley. She earned her M.S. degree in computer science from the University of California, Berkeley, and her B.S. in mathematics from the California Institute of Technology. She is also a former airshow and test pilot with over 5,000 cockpit hours.